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Large-Scale Yielding Failure Prediction of Notched Ductile Plates by Means of the Linear Elastic Notch Fracture Mechanics

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Прогнозирование разрушения при полномасштабном течении в пластинах с надрезом на основе линейной упругой механики разрушения

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Предложен критерий разрушения, основанный на линейной упругой механике разрушения, для прогнозирования инициирования трещины растяжения от затупленного V-образного надреза при большой пластичности вблизи надреза. Описаны некоторые современные экспериментальные результаты по разрушению при растяжении пластичных алюминиевых пластин с V-образным надрезом. С целью избежания сложного и трудоемкого упругопластического анализа для прогнозирования несущей способности алюминиевых пластин с V-образным надрезом используются концепция эквивалентного материала и критерий усредненной плотности энергии деформации, базирующийся на линейной упругой механике разрушения. Показано хорошее соответствие между экспериментальными данными и результатами прогнозирования на основе концепции эквивалентного материала и критерия усредненной плотности энергии деформации.

Ключевые слова: надрез, пластичный материал, зарождение трещины, понятие эквивалентного материала, критерий усредненной плотности энергии деформации, полномасштабное течение.

Introduction. In general, ductile metallic materials are divided into two main categories. The first and second categories involve those materials exhibiting negligible and considerable strain-hardening in the plastic zone, respectively. An accurate factor to realize the category is normally the difference between the yield and tensile strengths of the material. The fracture mechanics analyses for the two categories are really different. The

materials inside the first category have usually valid fracture toughness value in the context of the linear elastic fracture mechanics (LEFM); i.e., K-based fracture toughness K_{Ic} or K_c , where K_{Ic} is utilized in plane-strain conditions, which is usually satisfied for relatively large thicknesses, and K_c in plane-stress conditions, relating to small thicknesses. For ductile metallic materials inside the second category, a LEFM-based fracture toughness, K_{Ic} or K_c , is not obtained according to the standard code ASTM E399-12e3 [1], but instead, some other fracture toughness values in the context of the elastic-plastic fracture mechanics (EPFM), e.g., the crack tip opening displacement (CTOD), crack tip opening angle (CTOA), resistance curve (*R*-curve), critical *J*-integral, etc., are utilized to characterize the resistance of material against crack propagation [2, 3]. Despite valid K_{Ic} or K_c values for the first category, if the size of the plastic zone ahead of the crack tip at the propagation instance becomes considerable, particularly for small thicknesses, the effects of the plastic zone on the prediction of crack propagation should be considered in the fracture models.

Although the physical meaning of the notch fracture toughness (NFT) is completely different from that of the fracture toughness (FT); the NFT means the resistance of a notch against initiating crack(s) from its border, while the FT means that of a crack against propagation; the failure process in notched members is very similar to that in cracked ones. For example, in a V-notched ductile member, a plastic region initiates from the notch border during loading and grows till the onset of crack initiation from the notch border at which the notched member shows its load-carrying capacity (LCC). For predicting the LCC of a V-notched ductile component, various criteria exist depending on the existence of valid or invalid K-based fracture toughness value. For ductile materials inside the first category, the thickness of the notched component plays an important role in selecting appropriate failure criteria. If the thickness becomes larger than a specified value, which is usually computable, no considerable plastic zone will form around the notch at the crack initiation instance and hence, the failure criteria in the context of the linear elastic notch fracture mechanics (LENFM) can accurately be utilized for the LCC prediction. For small thicknesses (e.g., thin plates), however, the plastic zone size is significant; so considering the effects of plastic deformations around the notch in the LCC prediction is necessary, and using directly the LENFM-based criteria is not allowable.

Aluminum alloys are extensively utilized in engineering structures, particularly aerospace structures, due to their good strength/weight ratio, high toughness and ductility, good fatigue resistance and fracture toughness etc. Most of such alloys locate inside the first category of ductile materials and hence, they have valid values of K-based fracture toughness. Consequently, selecting the failure criterion for predicting the NFT for such alloys strongly depends on thickness. Recently, two papers have been published on the NFT of aluminum alloys [4, 5]. The mode I NFT of a commercial aluminum alloy containing U-notches of various tip radii has been evaluated by Vratnica et al. [4]. They have carried out the fracture experiments on 10 mm thick single-edge-notched-bend (SENB) specimens; measured the NFT, and predicted successfully the experimentally obtained NFT values by means of a stress-based criterion that employs the linear-elastic stress distributions around a blunt crack-like notch [4]. Madrazo et al. [5] have also performed a research work on mode I fracture of notched Al 7075-T651 specimens. They have conducted tensile fracture tests on the notched compact-tension (CT) specimens of 20 mm thick and experimentally measured the mode I NFT values [5]. Such values have then been well predicted by means of the linear elastic theory of critical distances (TCD) [5]. It is worth noting that the TCD has two well-known approaches, namely the point method and the line method [6-9] which both have been employed in [5] for theoretical fracture assessments. According to the requirements reported in [1] for a valid K_{Ic} test and considering the materials tensile and fracture toughness properties reported in [4, 5]; the minimum thicknesses for valid fracture toughness tests can simply be calculated to be equal to about 40 and 5 mm, respectively. Comparing theses values with the notched specimen thicknesses reported in [4, 5] indicates that the notched specimens of [4] do not satisfy plane-strain conditions, while those of [5] do so. Accordingly, considerable plastic zone has been formed around notches of [4] at the crack initiation instance and hence, ignoring such a zone in the theoretical failure predictions is questionable.

Most recently, due to wide application of thin aluminum plates in aero-structures, a few research works have been carried out on ductile failure analysis of thin notched aluminum plates under various loading conditions [10-15]. In [10-15], Torabi and co-researchers have performed numerous failure tests on thin rectangular plates weakened by blunt V- and U-notches, and made of the aluminum alloys Al 7075-T6 and Al 6061-T6 under pure mode I and mixed mode I/II loading conditions from which the cracking behavior of the plates has been studied; the failure regime, e.g., the moderate-scale yielding (MSY), large-scale yielding (LSY) etc., has been determined; and the LCC of the notched plates has been experimentally measured. As observed during the experiments, for such thin plates, the plastic zone around the notch at the crack initiation instance is considerable and hence, the failure criteria in the context of the LENFM could not be used for predicting the LCC of the notched plates. In order to avoid time-consuming and complex elastic-plastic failure analyses and to make the LENFM-based failure criteria usable for such ductile problems, the equivalent material concept (EMC), proposed originally by the first author [16–19], has been employed. The experimentally obtained LCCs have been successfully predicted by using some well-known stress-based brittle fracture criteria, namely the point-stress (PS), mean-stress (MS) and maximum tangential stress (MTS) criteria together with the EMC [10-15].

The most recent research on ductile failure of notched components has been performed by Torabi et al. [20]. In [20], the experimentally recorded tensile LCCs of the V-notched Al 7075-T6 plates reported in [10] have been well predicted by means of the EMC in conjunction the well-known energy-based brittle fracture criterion, namely the averaged strain energy density (ASED) criterion [21–27]. It has been found in [20] that the EMC-ASED criterion could work well on mode I ductile failure of notched aluminum plates *under the MSY failure regime*. In order to examine the success of the EMC-ASED criterion also *under the LSY failure regime*, it is attempted in the present research to predict the tensile LCCs of the V-notched Al 6061-T6 plates reported most recently in [14]. A very good agreement is shown to exist between the theoretical and experimental results, demonstrating that the combination of the EMC and the ASED criterion is successful, regardless of the size of plastic zone around the notch at crack initiation instance.

1. Ductile Fracture Test Results Reported in the Literature. Most recently, Torabi and Keshavarzian [14] have published a set of ductile fracture test results on blunt V-notched thin plates under pure mode I loading. The material investigated has been the aluminum alloy Al 6061-T6 with the chemical composition and mechanical properties presented in Tables 1 and 2, respectively [14]. Figure 1 shows the tensile engineering and true stress–strain curves of the material [14]. As seen in Fig. 1, Al 6061-T6 exhibits ductile behavior in tension with negligible strain-hardening in the plastic region.

Si	Fe	Cu	Mn	Mg	Zn	Ni	Cr	Pb	Sn	Ti
0.61	0.48	0.17	0.05	0.86	0.02	0.003	0.2	0.001	0.001	0.08
В	Cd	Bi	Ca	Р	Sb	V	Zr	Со	Li	Al
0.004	0.001	0	0.001	0.002	0.008	0.025	0.0002	0.003	0.001	97.6

Al 6061-T6 Chemical Composition [14] (in wt.%)

Table 1

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Mechanical property	Value		
Elastic modulus E, GPa	67		
Poisson's ratio	0.33		
Tensile yield strength (MPa)	276		
Ultimate tensile strength (MPa)	292		
Elongation at break (%)	11		
Engineering strain at maximum load	0.034		
True fracture stress (MPa)	299		
Fracture toughness K_c , MPa \sqrt{m}	38		
Strain-hardening coefficient (MPa)	314		
Strain-hardening exponent	0.021		

Some of the Mechanical Properties of Al 6061-T6 [14]

Table 2



Fig. 1. Tensile engineering and true stress-strain curves of Al 6061-T6 [14].

The specimen utilized in [14] to perform the fracture experiments has been a rectangular plate of 4 mm thick containing a centrally located rhombic slit with four round-tip V-notches; two of which are subjected to pure mode I loading (i.e., the opening mode) as a result of a tensile load remotely applied to the specimen. A scheme of the specimen is depicted in Fig. 2, including the geometrical and loading parameters [14].

The parameters 2α , ρ , 2a, L, W, and P in Fig. 2 are the notch angle, the notch radius, twice the notch length (i.e., the horizontal diameter of the rhombic slit), the specimen length, the specimen width, and the applied tensile load, respectively. The specimen dimensions have been considered in the fracture tests to be as follows: $2\alpha = 30$, 60, and 90°, $\rho = 1, 2$, and 4 mm, 2a = 25 mm, L = 160 mm, and W = 50 mm [14]. Totally, twenty seven displacement-controlled monotonic fracture tests have been conducted in [14].

Three experimentally obtained evidences have been reported in [14], demonstrating that the V-notched Al 6061-T6 plates have failed by the LSY regime. Note that the failure has been defined in [14] to be the crack initiation from the notch tip. The evidences are: (i) permanent deformations of the specimen ligament in large amount at the crack initiation instance (it could be captured by naked eye); (ii) considerable opening of the notch mouth

before failure, and (iii) large nonlinear portion in the load–displacement curve before the peak. To numerically confirm the experimentally observed LSY failure of the V-notched Al 6061-T6 plates, some elastic-plastic finite element (FE) analyses have been performed on the plates, and it has been shown that a large portion of the ligament (more than 50%) experiences plastic deformations at the onset of crack initiation from the notch tip, proving well the LSY failure regime [14].

The experimental values of the LCC of the tested specimens reported in [14] are summarized in Table 3 for various notch angles and notch radii. In Table 3, the parameter P_i (i = 1, 2, 3) denotes the LCC in the three repeated tests and P_{av} is the average of the three LCCs.

Table 3

2α , deg	ρ, mm	<i>P</i> ₁ , N	P_2 , N	<i>P</i> ₃ , N	P_{av} , N	
30	1	30,567	30,350	29,997	30,305	
	2	31,674	31,884	31,913	31,824	
	4	31,862	31,615	31,652	31,710	
60	1	31,325	31,685	31,876	31,629	
	2	31,668	31,538	31,711	31,639	
	4	32,119	31,869	31,946	31,978	
90	1	29,334	30,626	30,114	30,025	
	2	31,501	31,468	31,485	31,485	
	4	31,516	31,398	31,450	31,455	

Experimental LCCs of the V-Notched Al 6061-T6 Specimens for Various Notch Angles and Notch Radii [14]





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In the next section, the EMC is briefly described, capable of equating the real ductile material having elastic-plastic behavior with a virtual brittle material exhibiting linear-elastic behavior.

2. Equivalent Material Concept. The EMC has been originally proposed by Torabi [16] with the aim to take a permission to analyze ductile failure of notched members by means of the LENFM principles. Two main mechanical properties, namely the fracture toughness and the tensile strength are necessary for failure analysis of notched components in the context of the LENFM. For the first property, it is assumed that the ductile material showing considerable plastic zone in the tensile stress-strain curve has valid *K*-based fracture toughness (K_{Ic} or K_c) value. For the second property, the ultimate tensile strength of the ductile material cannot directly be used in the LENFM-based fracture criteria, since the material exhibits considerable nonlinear behavior in the plastic region. To overcome this restriction, the EMC is utilized in which the ductile material is equated with a virtual brittle material with linear elastic behavior; both absorb the same amount of the tensile strain energy density (SED) for the crack initiation to take place. By computing the SED for ductile material till the ultimate point and setting it equal to that for the virtual brittle material till final fracture, the tensile strength of the equivalent material can be computed.

Considering the power-law relationship between the stress and the plastic strain in the plastic region of the ductile material, an expression has been reported in a few recent references for computing the tensile strength of the equivalent material, σ_f^* . It is [10–15]:

$$\sigma_f^* = \sqrt{\sigma_Y^2 + \frac{2EK}{n+1} [\varepsilon_{u,true}^{n+1} - (0.002)^{n+1}]},$$
(1)

where the parameters σ_f^* , σ_Y , *E*, *K*, *n*, and $\varepsilon_{u,true}$ are the tensile strength of the equivalent material, yield strength of the real ductile material (e.g., Al 6061-T6 in [14]), elastic modulus, strain-hardening coefficient, strain-hardening exponent, and the true plastic strain at maximum load, respectively. Simply, $\varepsilon_{u,true}$ can be calculated by using Eq. (2), in which the parameter ε_u denotes the engineering plastic strain at maximum load:

$$\varepsilon_{u,true} = \ln(1 + \varepsilon_u). \tag{2}$$

Now, the K_{Ic} (or K_c) and σ_f^* , which are the two main fracture properties of the equivalent material, can simply be utilized in various fracture criteria in the field of the LENFM, e.g., the ASED criterion, for predicting the onset of crack initiation from the notch tip in ductile materials. In the forthcoming sections, the ASED criterion is briefly described and by linking it to the EMC [i.e., using K_{Ic} (or K_c) and σ_f^* values in the criterion], the tensile LCCs of the V-notched Al 6061-T6 plates (see Table 3) that fail by LSY regime are theoretically predicted.

3. Brief Description of the ASED Criterion. The most important point for designers is certainly the existence of appropriate failure models to predict the load-carrying capacity of components weakened by notches. With the aim to provide such models, a strain energy density based criterion has been proposed by Lazzarin and co-authors [21, 22], by which the experimental fracture loads of notched specimens can properly be estimated.

The strain energy density factor S was defined for sharp cracks by Sih [28] as the product of the strain energy density by a specified critical distance measured from the crack tip. Fracture was thought of as controlled by a critical value S_c , whereas the crack growth direction was determined by imposing a minimum condition on the factor S.

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The method proposed by Sih is a point-wise criterion, while the ASED approach as presented in [21, 22] suggests that brittle fracture takes place when the strain energy density averaged over a known control volume is equal to a critical value W_c . This value varies from material to material but it is independent of the notch geometry. The control volume is thought of as dependent on the ultimate tensile strength and on the fracture toughness K_{Ic} in the case of brittle or quasi-brittle materials subjected to static and monotonic loads. Such a method was formalized and applied first to sharp V-notches under mode I and mixed mode I/II loadings [21] and later extended to blunt U- and V-notches [22, 29], also under multiaxial I/III static loadings [24, 25]. Some recent developments and applications are reported in [22, 30–32], with some considerations also to three-dimensional effects [33–35], which have been widely discussed in [36].

For sharp cracks, the control volume is a circle of radius R_c centered at the crack tip (Fig. 3a). Under plane-strain conditions, the critical length, R_c , can be evaluated by the following expression [37]:

$$R_{c} = \frac{(1+\nu)(5-8\nu)}{4\pi} \left(\frac{K_{\rm Lc}}{\sigma_{t}}\right)^{2}.$$
(3)

In Eq. (3), K_{Ic} is the fracture toughness, ν is Poisson's ratio, and σ_t is the ultimate tensile strength of the material.



Fig. 3. Control volume (area) for sharp crack (a), sharp V-notch (b) and blunt V-notch (c) under mode I loading. Distance $r_0 = \rho(\pi - 2\alpha)/(2\pi - 2\alpha)$.

For a sharp V-notch, the control volume becomes a circular sector of radius R_c centered at the notch tip (Fig. 3b), while for a blunt V-notch under mode I loading, the volume assumes the crescent shape, as shown in Fig. 3c, where R_c is the depth measured along the notch bisector line. The outer radius of the crescent shape is equal to $R_c + r_0$, being r_0 the distance between the notch tip and the origin of the local coordinate system (see Fig. 3c). Such a distance depends on the V-notch opening angle 2α , according to the expression $r_0 = \rho(\pi - 2\alpha)/(2\pi - 2\alpha)$. For more details see also [22].

4. Application of EMC in Combination with ASED Criterion. The ASED approach is applied here by considering the mechanical properties of the equivalent material. The critical strain energy density, $W_{c,EMC}$, is evaluated by using Eq. (4) and considering $\sigma_f^* = 1066$ MPa, as obtained by applying the EMC according to Eq. (1):

$$W_{c,EMC} = \frac{(\sigma_f^*)^2}{2E}.$$
 (4)

The critical strain energy density results to be equal to 8.48 MJ/m³. The control radius R_c is evaluated by using Eq. (3) with $\sigma_t = \sigma_f^* = 1066$ MPa. It has been found $R_c = 0.317$ mm.

The ASED occurring inside the control volume embracing the edges of V-notches, \overline{W} , is calculated numerically by using the FE code ANSYS[®]. For each geometry, a model is created which requires an accurate definition of the control volume where the ASED should be averaged (see Fig. 3c). All FE analyses are performed by using second order eight-node plane elements (PLANE 183 of the ANSYS[®] element library) under plane-strain conditions. A detail of the FE mesh adopted inside the control volume is reported in Fig. 4. In particular, typical refined and coarse FE meshes are shown in Fig. 4a and 4b, respectively. As it is well-known from [30–32, 38, 39], the ASED can correctly be evaluated also with very coarse FE meshes and, for this reason, all numerical models are performed here using the FE mesh shown in Fig. 4b.



Fig. 4. Example of the refined FE mesh employed in the control volume (a). The same results can be achieved by using a coarse FE mesh, as that shown in (b).

5. **Results and Discussion**. Table 4 summarizes the outlines of the experimental, numerical and theoretical findings for V-notched specimens with three different notch radii $(\rho = 1, 2, \text{ and } 4 \text{ mm})$ and opening angles $(2\alpha = 30, 60, \text{ and } 90^\circ)$, analyzed by means of the ASED approach. In particular, Table 4 reports the experimental loads to failure (*P*) for all notch radii ρ and opening angles 2α compared with the theoretical values (P_{ASED}), obtained on the basis of the ASED evaluation. P_{ASED} is the theoretical load obtained by keeping a constant averaged strain energy density (\overline{W}) equal to 8.48 MJ/m³, that is the material critical value, over the control volume.

The last columns of the table present the deviations between the values of the experimental failure loads and the theoretical ones evaluated by means of ASED criterion. Δ is defined as the ratio between the experimental load and the theoretical one for each case.

It is clearly seen in Table 4 that the great majority of predictions are well inside the scatter of $\pm 20\%$, with some results inside the scatter $\pm 10\%$. Only few exceptions are characterized by a deviation between 20 and 25%.

A synthesis in terms of the square root value of the local strain energy averaged over the control volume of radius R_c (namely, \overline{W}), normalized with respect to the critical energy of the equivalent material, $W_{c,EMC}$, as a function of the notch opening angle 2α is shown in Fig. 5. The plotted parameter is proportional to the failure load. The aim is to investigate the influence of the notch geometry on the fracture prediction based on the ASED approach. Also from the graphical point of view, it is obvious that most of values fall inside a scatter ranging from 0.80 to 1.20 with the majority of the data inside 0.90 to 1.10. Only few exceptions are outside the scatter band, having a normalized ASED between

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Table 4

Critical Loads Predicted by Means of ASED Criterion in Combination with EMC								
2α , deg	$ ho, { m mm}$	<i>P</i> ₁ , N	P_2 , N	<i>P</i> ₃ , N	P_{ASED} , N	Δ_1	Δ_2	Δ_3
30	1	30,567	30,350	29,997	31,711	0.96	0.96	0.95
	2	31,674	31,884	31,913	35,951	0.88	0.89	0.89
	4	31,862	31,615	31,652	42,518	0.75	0.74	0.74
60	1	31,325	31,685	31,876	29,943	1.05	1.06	1.06
	2	31,668	31,538	31,711	34,349	0.92	0.92	0.92
	4	32,119	31,869	31,946	41,092	0.78	0.78	0.78
90	1	29,334	30,626	30,114	28,559	1.03	1.07	1.05
	2	31,501	31,468	31,485	32,465	0.97	0.97	0.97
	4	31,516	31,398	31,450	39,009	0.81	0.80	0.81



Fig. 5. Synthesis of fracture data in terms of normalized ASED.

0.75 and 0.80. The synthesis confirms also the choice of the control volume, which seems to be suitable to characterize the material behavior under pure mode I loading. The scatter of the experimental data presented here is in very good agreement with the recent database in terms of ASED reported in a recent review of the approach, which dealt with brittle and quasi-brittle materials [22].

Conclusions. The present research is aimed to predict tensile crack initiation in Al 6061-T6 thin plates weakened by blunt V-notches. Some experimental results relevant to tensile failure of V-notched ductile aluminum plates, published in a recent contribution, were briefly described. In that work, on the basis of experimental observations, a large-scale yielding failure regime was demonstrated for the considered aluminum plates. With the aim of theoretically predicting the experimental results, the equivalent material concept was adopted in conjunction with the averaged strain energy density evaluated over a control volume. Without requiring elastic-plastic finite element analyses, it was shown

that the combination of the averaged strain energy density approach and the equivalent material concept can successfully predict the load-carrying capacity of the V-notched Al 6061-T6 plates, characterized by large plastic deformations ahead of the notch tip at the onset of crack initiation.

Резюме

Запропоновано критерій руйнування, що базується на лінійній пружній механіці руйнування, для прогнозування ініціювання тріщини розтягу від затупленого V-подібного надрізу при значній пластичності поблизу надрізу. Описано деякі сучасні експериментальні результати щодо руйнування при розтязі пластичних алюмінієвих пластин із V-подібним надрізом. Із метою запобігання складного і трудомісткого аналізу для прогнозування несівної здатності алюмінієвих пластин із V-подібним надрізом використовуються концепція еквівалентного матеріалу і критерій усередненої щільності енергії деформації, що базується на лінійній пружній механіці руйнування. Показано хороший збіг експериментальних даних із результатами прогнозування на основі концепції еквівалентного матеріалу і критерію усередненої щільності енергії.

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